

親機子機分離機構を備えた月惑星探査機着陸機構の2次元シミュレーション

Two-Dimensional Simulation of Lunar/Planetary Exploration Spacecraft Landing Mechanism Using BESM

原 進 (名大) ○石川 凌 (名大) 大槻 真嗣 (JAXA) 渡辺 翼 (名大)
佐伯 直亮 (名大)

Susumu HARA*, Ryo ISHIKAWA*, Masatsugu OTSUKI**, Tsubasa WATANABE*
and Naoaki SAEKI*
*Nagoya University, **ISAS/JAXA



This study discusses a lunar/planetary spacecraft landing mechanism using energy conversion. A part of the authors has already proposed Base-Extension Separation landing Mechanism (BESM) and its effectiveness was confirmed in one-dimensional simulations and experiments. This study shows two-dimensional response analysis using BESM. The effectiveness of BESM for falling to slopes is verified.

Background

- ◆ Explorations of the lunar attract attention

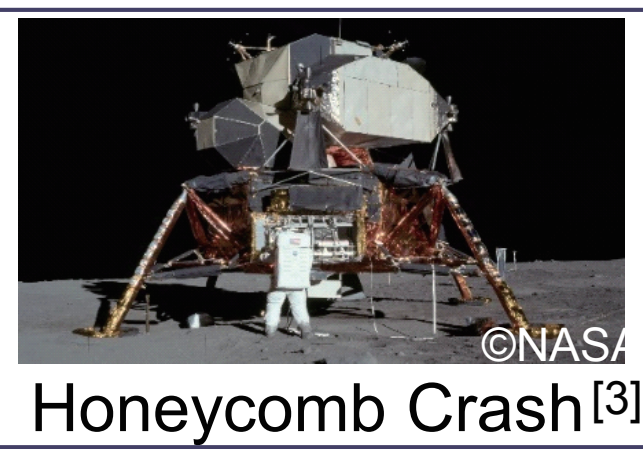
Requirement

To realize soft landing on severe regions...
for example **slopes** or **steps**.

- ◆ Previous methods and their problems



Airbag[2]



Honeycomb Crash[3]

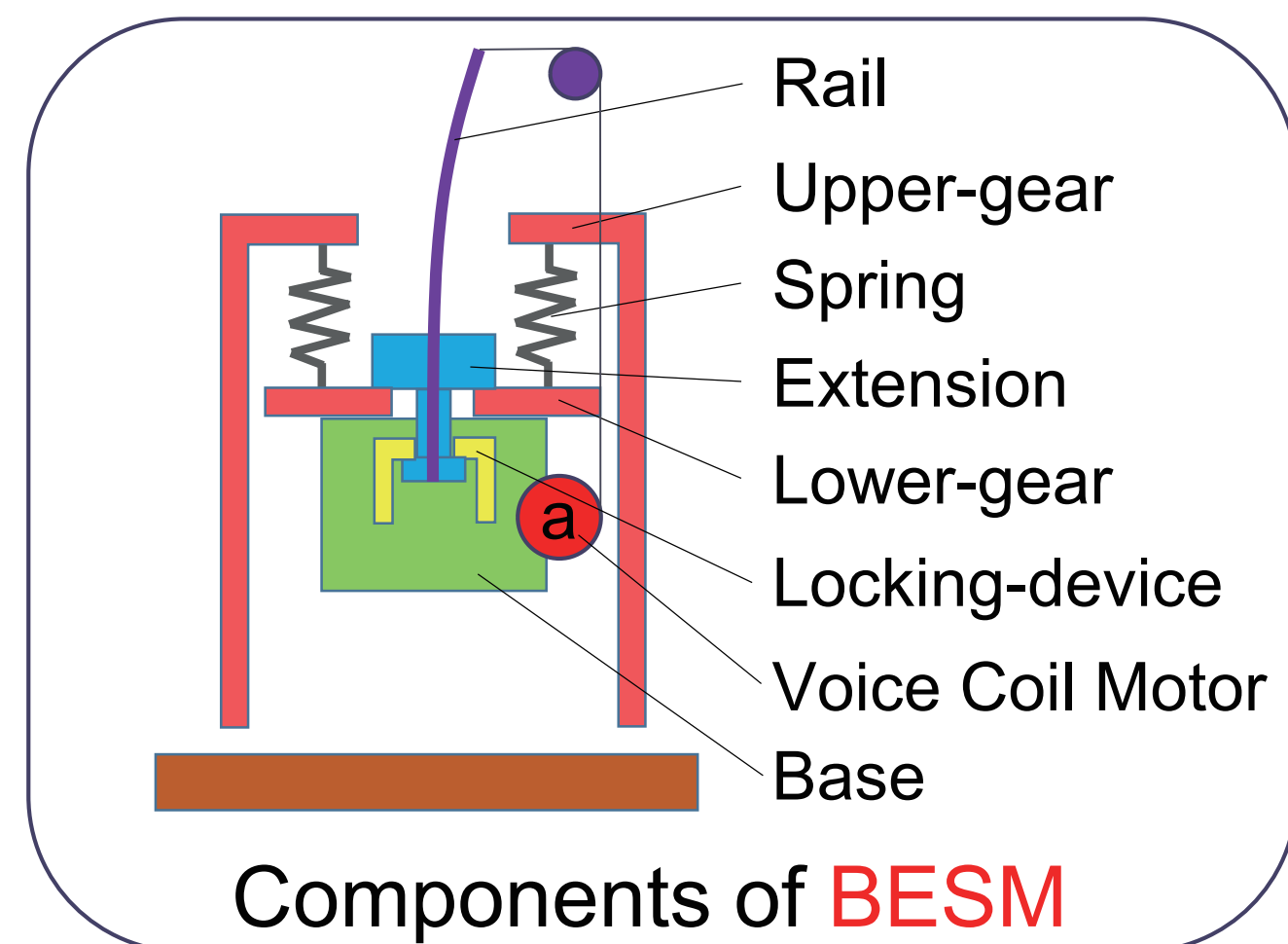


Sky Crane[4]

Problems
High rebound
Impossibility to reuse
Complex control

BESM

Base-Extension Separation landing Mechanism

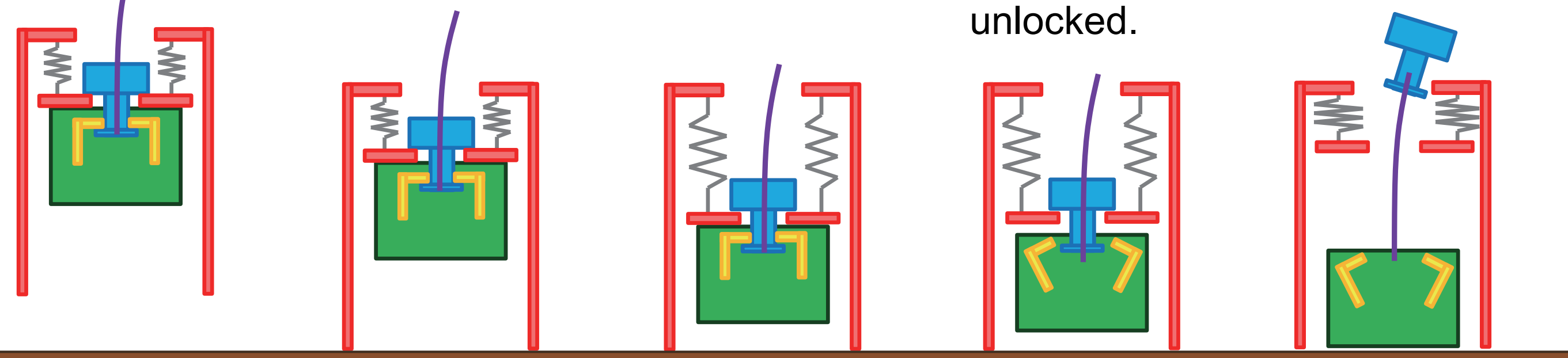


Components of BESM

BESM can realize...

- Low rebound
- Reuse
- Passive landing

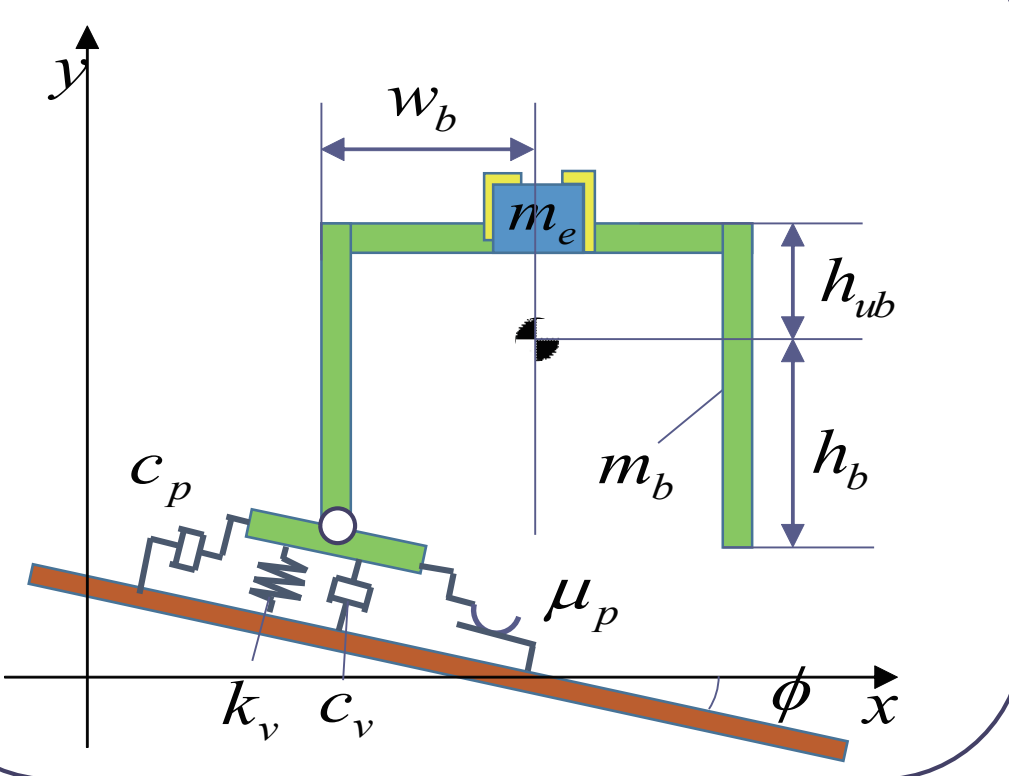
1. Falling
2. Landing
3. Spring is stretched and velocity of the base is reduced.
4. When the springs reach stroke length, locking-device is unlocked.
5. Extension is launched. Base lands softly.



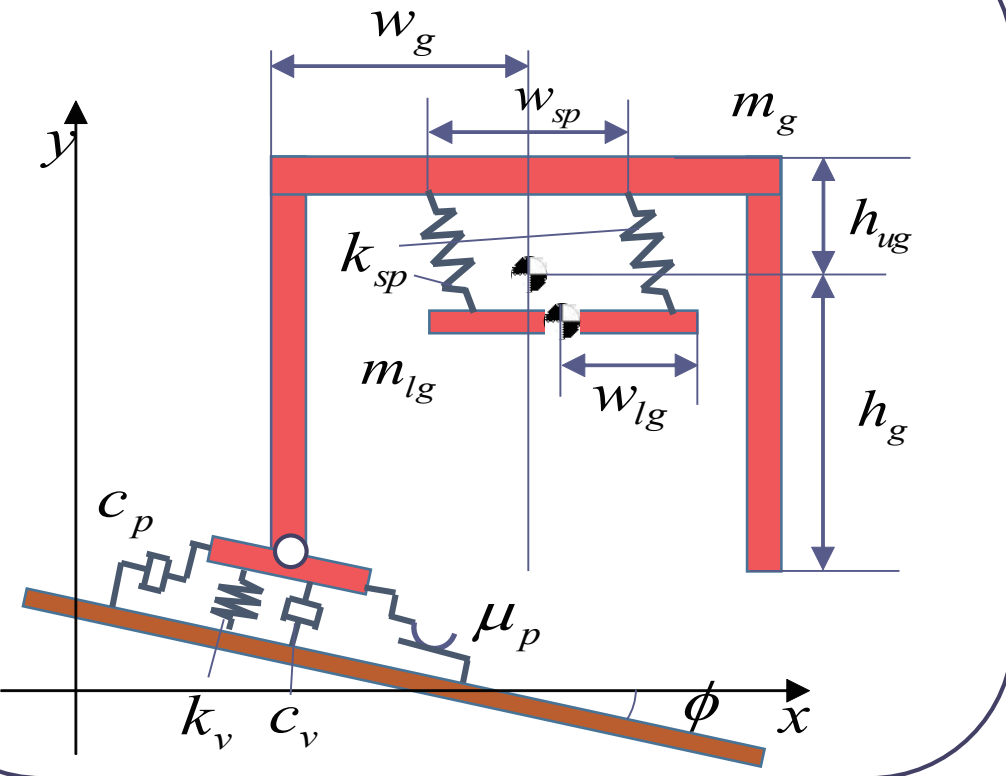
• The V. C. M. is omitted.

Models for simulation

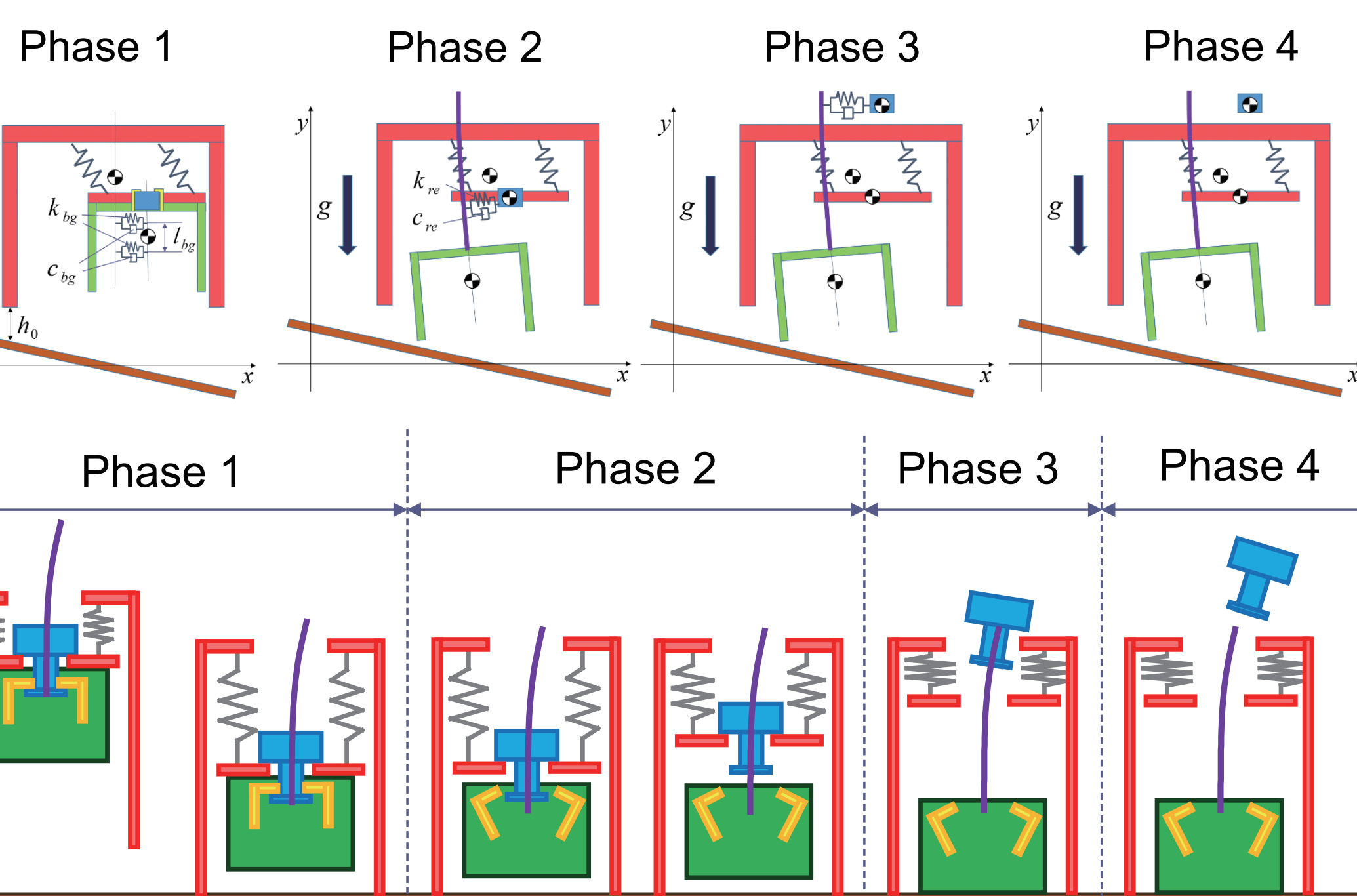
- Model for the base



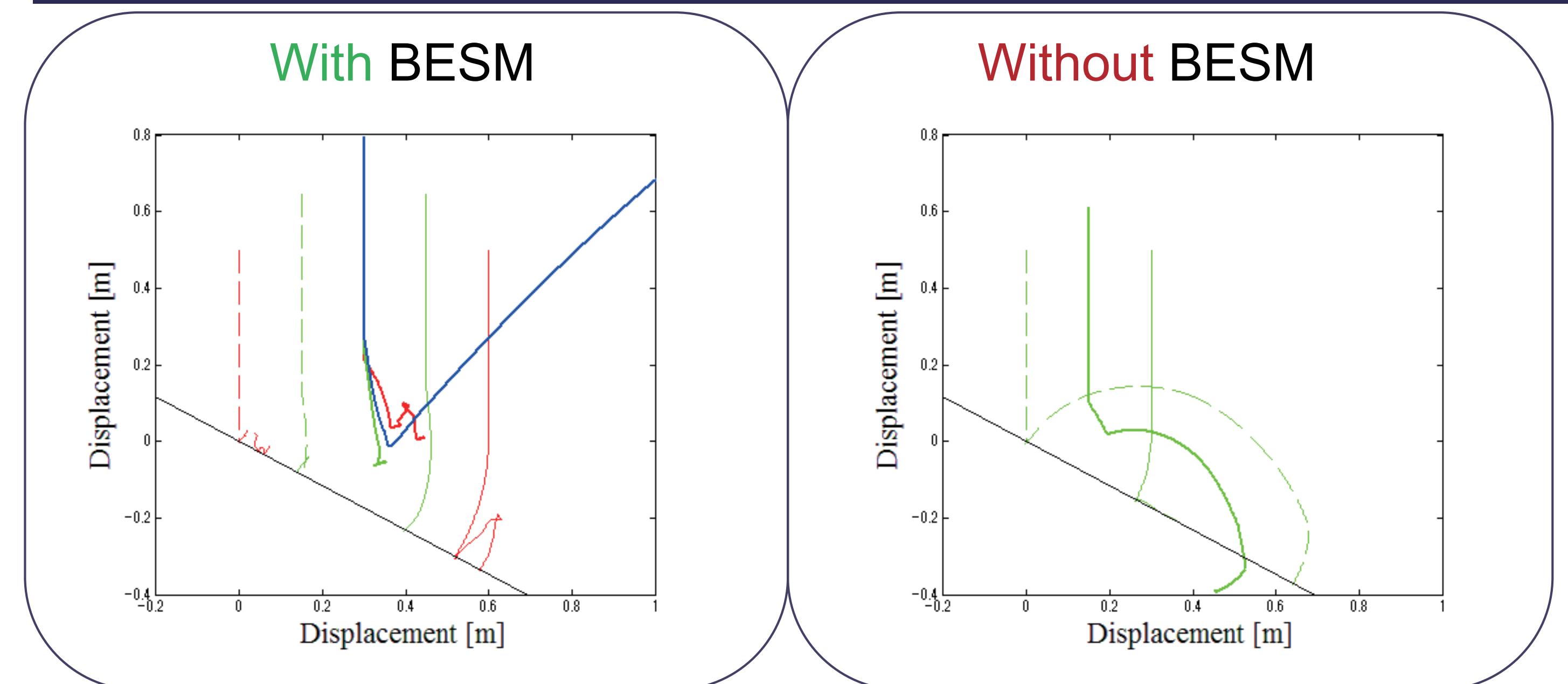
- Model for the gear



m_b	Base mass [kg]	1.9
m_e	Extension mass [kg]	0.1
w_b	Width of the base [m]	0.15
h_b	Base lower length [m]	0.11
h_{ub}	Base upper length [m]	0.04
m_g	Gear mass [kg]	0.4
w_g	Width of the gear [m]	0.3
h_g	Gear lower length [m]	0.216
h_{ug}	Gear upper length [m]	0.108
m_{lg}	Lower gear mass [kg]	0.05
w_{lg}	Width of the lower gear [m]	0.15
k_{sp}	Stiffness of springs [N/m]	220
w_{sp}	Width between springs [m]	0.15
l_{st}	Stroke length [m]	0.16
k_v	Vertical stiffness of ground [N/m]	1667
c_v	Vertical damping of ground [N·s/m]	556
c_p	Parallel damping of ground [N·s/m]	556
μ_p	Parallel dynamic coefficient of friction of ground [-]	0.8
ϕ	Ground angle [°]	-30
k_{bg}	Stiffness of the restriction between the base and gear [N/m]	1.0×10^4
c_{bg}	Damping of the restriction between the base and gear [N·s/m]	100
l_{bg}	length between the two couples of spring and damper [m]	0.08
k_{re}	Stiffness of the restriction between the extension and rail [N/m]	1.0×10^4
c_{re}	Damping of the restriction between the extension and rail [N·s/m]	100
h_0	Initial falling height [m]	0.5
g	Acceleration of gravity [m/s ²]	9.8



Simulation results

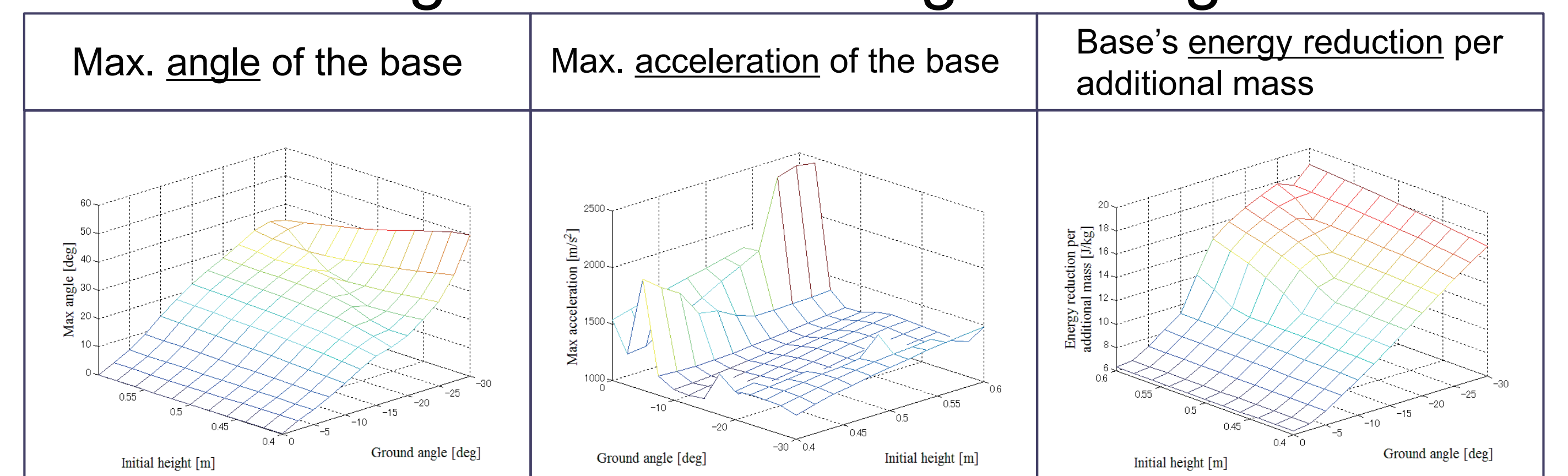


Red: gear Green: base Blue: extension
Bold: center of gravity Thin: right tip Dashed: left tip

- 30° の着地面への着陸において、親機の転倒を防ぐことができた。

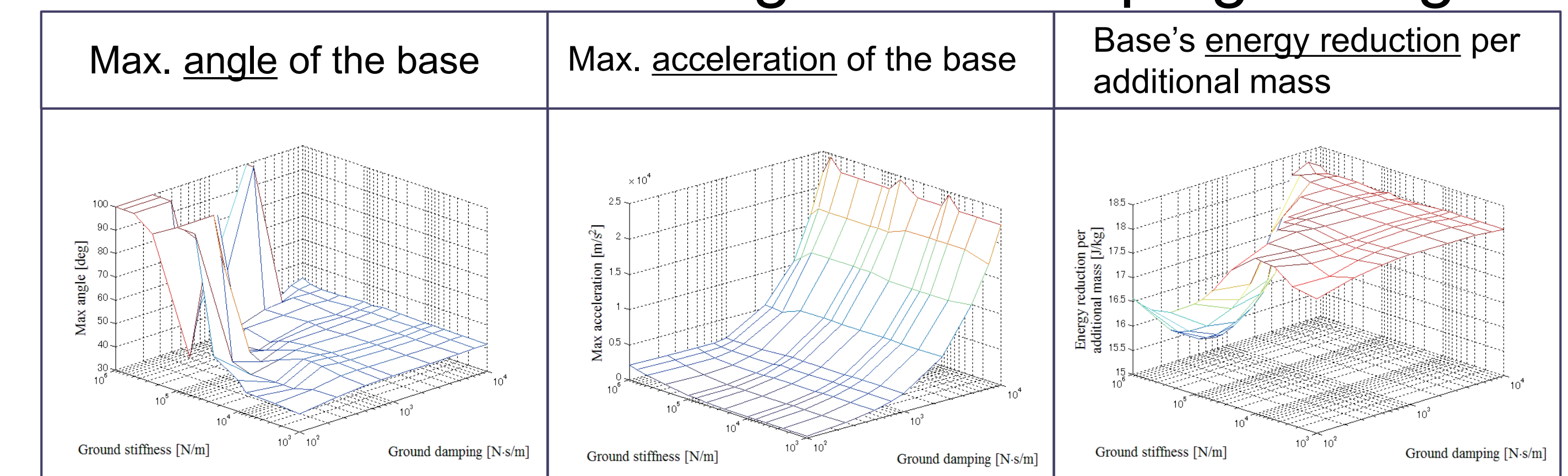
- BESM **can prevent** from tipping for falling to 30° slope.

- ◆ Ground angle and initial height change.



- 全ての条件で、親機の転倒を防ぐことができた。
- BESM can prevent from tipping under **all conditions**.
- Acceleration becomes high under small ground angle conditions.
- Energy reduction becomes large under large ground angle conditions.

- ◆ Ground stiffness and ground damping change.



- 一部を除くほぼ全ての条件で、親機の転倒を防ぐことができた。
- BESM can prevent from tipping under **most of conditions**.
- Acceleration almost depends on **only** ground damping.
- Efficiency to prevent tipping may relate to efficiency of energy reduction.

Conclusion

- BESM has robustness for change of ground conditions and initial height.
- Efficiency to prevent from tipping may relate to efficiency of energy reduction.
- Acceleration of the base becomes high under conditions that ground angle is small.

References

- [1] Saeki, N., Hara, S., Otsuki, M., Watanabe, T., and Yamada, Y., Base-Extension Separation Mechanism for Planetary Exploration Spacecraft Landing, *Proceedings of the 29th International Symposium on Space Technology and Science (29th ISTS)*, (2013), 2013-d-27.
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- [3] Rogers, W. F., Apollo Experience Report - Lunar Module Landing Gear Subsystem, Technical Report TND-6850, NASA (1972).
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- [5] 横山隆明, 月惑星砂地盤への着陸衝撃力算定手法に関する研究, 総合研究大学院大学博士論文, (2008), pp. 26-36.